

Autorotation procedure

- Collective reduce, aft cyclic to get nose up
- Set IAS to Vy
- Twist grip shut off detent
- Maneuver aircraft into wind
- At 70ft AGL cyclic flare
- At 20ft collective increase and cyclic forward
- Collective increase to cushion landing

Engine relighting

- When Ng < 10%
- Normal engine start procedure
- Minimum 1000ft AGL required

GOV failure

NR drop:

- Collective reduce to maintain NR in green arc
- Twist grip → check in flight position
- If necessary apply autorotation

NR increase:

- Collective increase to maintain NR in green arc
- Twist grip slightly reduce
- Land as soon as possible
- Initiate a shallow approach
- Set Torque at around 30%
- Set ground speed below 10kts
- After touch down reduce twist grip to idle before lowering collective

Smoke in the cabin

Source of smoke identified:

- Corresponding system OFF
- Ventilate the cabin

Source of smoke not identified:

- Heating/demisting OFF
- Battery and Generator OFF
- Ventilate the cabin
- All consumers OFF
- Battery ON → check DC voltage
- Generator ON → check DC voltage

If DC parameters faulty:

- Generator OFF
- Unnecessary equipment OFF
- Land as soon as practicable

If DC parameters correct and no smoke detected:

- All consumers one by one ON to identify failed system then keep it off
- Continue flight upon equipment failed

Flight control servo jam

- Maintain attitude
- HYDR switch OFF
- Set IAS to Vy → apply HYDR failure procedure

VEMD failure

One screen failure:

- Read all information on other screen
- Use scroll on display or collective if necessary

Both screen failure:

- Check battery and generator ON
- Set IAS to max 100 kts (-2kts/1000ft)
- Carry out a no hover landing

Loss of TR thrust in flight

-Indicated by nose left yaw, cannot be stopped by right pedal

Hover IGE or within H/V:

-Twist grip → IDLE
-Collective → INCREASE to cushion landing

Hover OGE:

-Collective → REDUCE depending on available height
-Cyclic → FORWARD to gain airspeed
-Airspeed → MAINTAIN Vy or higher
-Collective → ADJUST to obtain minimum sideslip angle
-Depending on minimum speed, LAND or AR to hard surface

In CRUISE:

-Airspeed → Maintain Vy or higher
-Collective → ADJUST to obtain minimum sideslip angle
-Depending on minimum speed, LAND or AR to hard surface

HYD system failure

-Adjust airspeed for comfortable control
-HYD switch → verify ON
-If HYD not restored → HYD switch OFF
-Land as soon as practical

Abnormal NR/NF indications

NR indication failure:

-Collective → Maintain Tq > 10%
-NR is given by NF pointer
-Continue flight

NF indication failure:

-Check in normal operating range with Tq > 0%
-Continue flight

ENGINE OIL > 110° C

-IAS set to Vy
-If temperature decreases → land as soon as practicable
-Otherwise → land as soon as possible
-Check oil pressure gauge
-If gauge confirms pressure loss, land immediately

Loss of OAT, Ng, Tq, T4

-FLI is replaced by 3-data symbology
-Continue flight

OAT indicator failure:

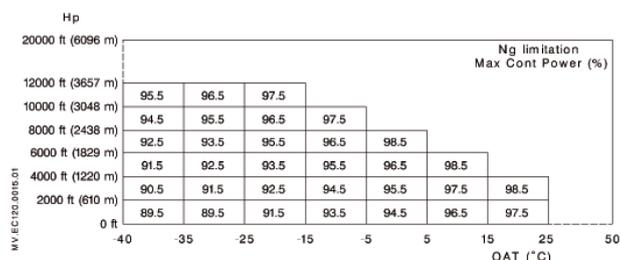
-Max T/O PWR Ng: 100%
-Max cont. PWR Ng: 98.5%

Ng indicator failure:

-If OAT > -10°C: T4 limited to 760°C
-If OAT ≤ -10°C: T4 limited to 750° C
-The T4 limitations displayed are the starting limitations

Tq meter failure:

-Comply with the table:



T4 indication failure:

-Comply with Ng and Tq limitations
-Do not start the engine

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|-----------------------------|---|
| ENG FIRE | -Indicates fire in engine compartment → procedure -land immediately |
| ENG P | -Check oil pressure gauge -If normal → Land as soon as practicable -If low or NIL → land immediately → perform AR |
| MGP P | -Excessive temp or low oil pressure of MRGB -Tq → set < 45% -Land as soon as possible (max. 30min flight time) |
| TWT GRIP | -Twist Grip → Open to FLIGHT position |
| HYDR | -Reduce Collective -Set IAS to Vy -HYD switch → OFF -Land as soon as possible -Shallow approach with slow running landing |
| BATT TEMP | -Battery → OFF -Check GEN voltage: |
| Voltage correct: | -Check BATT Temp, if decreases → Continue flight -If steady → Land as soon as practicable |
| Voltage > 32V | -BATT switch → ON -GENE switch → OFF -Unnecessary equipment OFF |
| GEN 32.0 V | -GEN → OFF -CWP → Monitor for BATT TEMP -Apply GENE procedure case A |
| GEN 30.0 V | -GEN voltage → MONITOR |
| GEN 24.0 V | - GENE procedure → apply -GEN voltage → Check -If > 24V → continue flight -If < 24V → apply GENE procedure case A |
| GEN 210 A | -Unnecessary equipment → OFF → Continue flight |
| GENE | -GEN voltage on VEMD → CHECK -Verify GENE switch → ON |
| Case A; GENE light rem. on: | -Push ELEC RST switch → if light out, apply Case B -Unnecessary equipment → OFF -BAT voltage on VEMD → Monitor -Land as soon as practicable -When battery voltage <18V, NR audio alarm will come on |
| Case B; GENE light out: | -Continue flight |

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|------------------|--|
| ENG CHIP | -Metal particles in engine oil circuit -Reduce Power → Land as soon as possible |
| MGB TEMP | -IAS → set to Vy -Monitor CWP -MGB Temp remains ON: land as soon as possible -Otherwise: land as soon as practicable |
| GB CHIP | -Metal particles in engine MGB or TGB oil circuit -IAS → set to Vy -Land as soon as practicable |
| BATT | -Battery is offline -Battery check ON -ELECT RST → PUSH -land as soon as practicable |
| BATT FUSE | -Battery fuse has blown → Battery is offline -Land as soon as practicable |
| FUEL | -Fuel quantity < 30kg -15min of flight time remaining with MCP -Avoid sideslip over 15° -Land as soon as possible |
| FUEL P | -Indicates low fuel pressure -At engine start up → FUEL PUMP ON -Reduce power -FUEL PUMP ON -Land as soon as possible |
| FUEL FILT | -Fuel filter clogged -Reduce power -Light remains ON → Land as soon as possible -Light OFF → Land as soon as practicable -In both cases: if Ng oscillations occur → Check NR -If NR normal → Land as soon as possible -Otherwise apply GOV FAILURE procedure |
| PITOT | -PITOT check ON |
| HORN | -HORN check ON |
| P2 TEMP | -Maximum temperature in heating duct exceeded -Check that air flows and air outlets not obstructed |